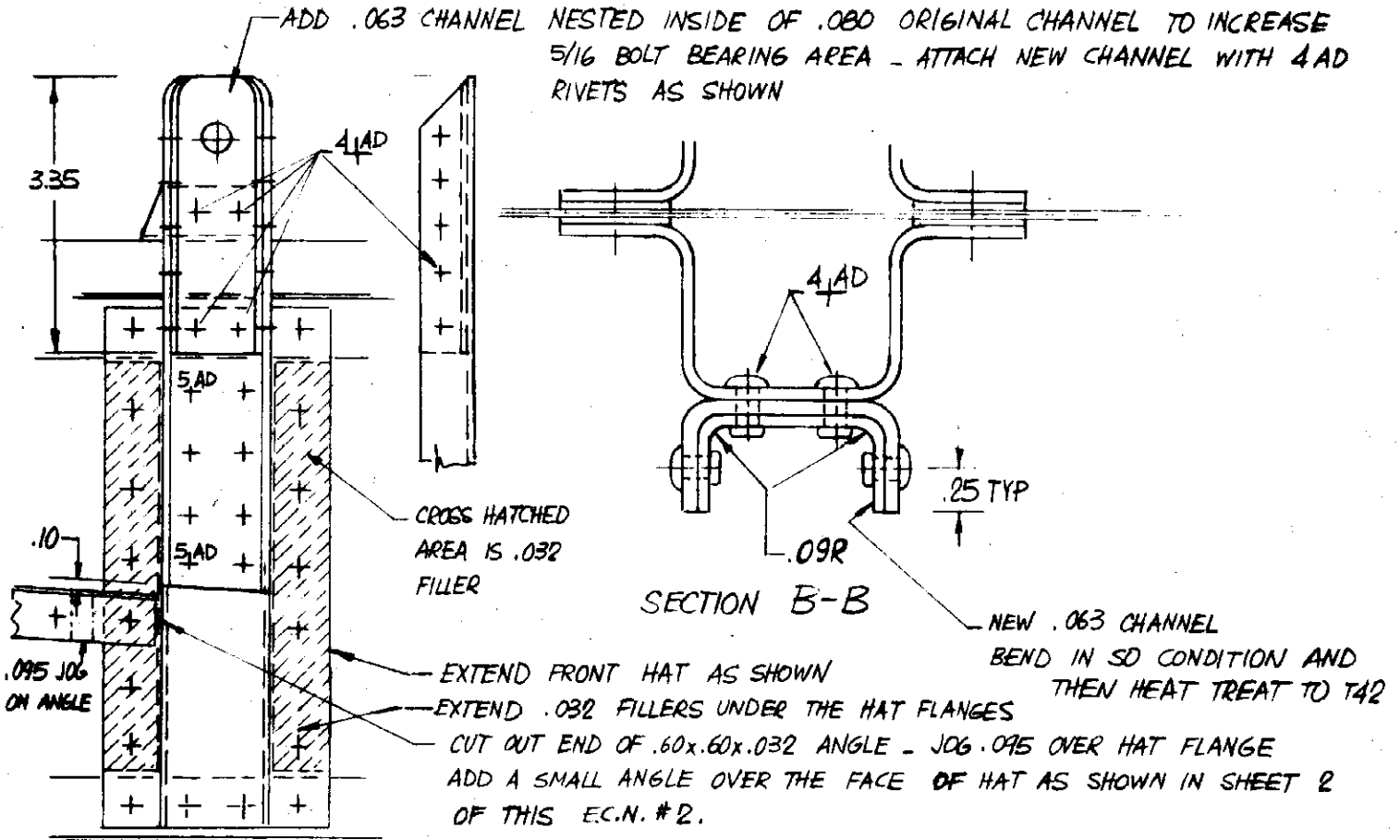


ENGINEERING CHANGE NOTICE # 2.

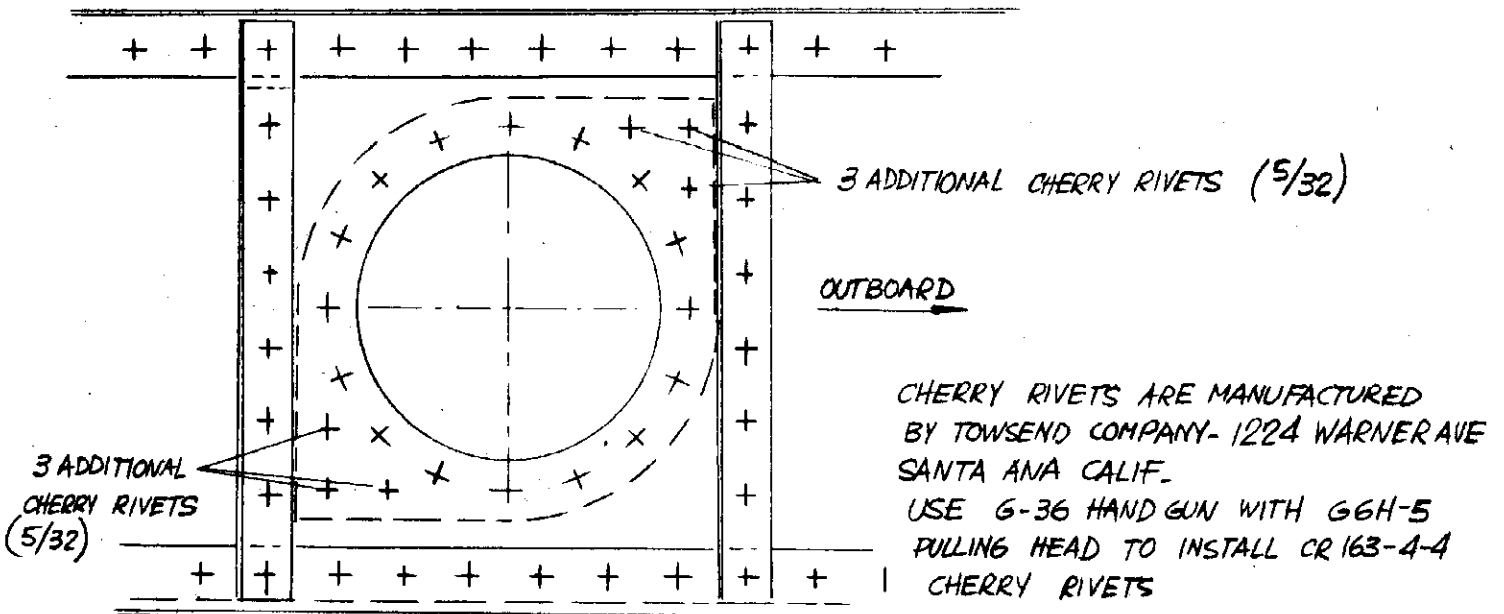
NOV - 1965

SHEET #1

DRAWING 1-10.002 - MAIN SPAR - WING (MANDATORY CHANGE)



DRAWING 1-10.002 - (MANDATORY CHANGE)



COVERS FOR ACCES HOLES TYP. 5 PLACES
 EXTEND CORNERS OF .050 COVERS AS SHOWN
 USE 5/32 BLIND CHERRY RIVETS (CR 163-4-4)
 TO ATTACH COVER TO WEB AFTER RIVETING LEADING EDGE SKIN

ENGINEERING CHANGE NOTICE #2

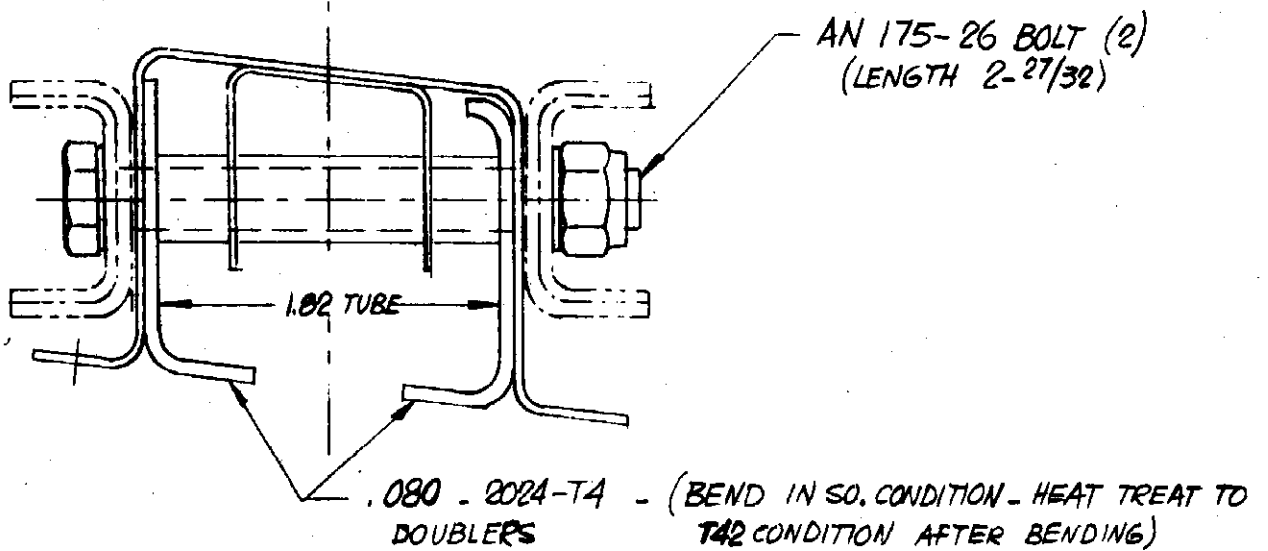
NOV. 1965

- SHEET #2

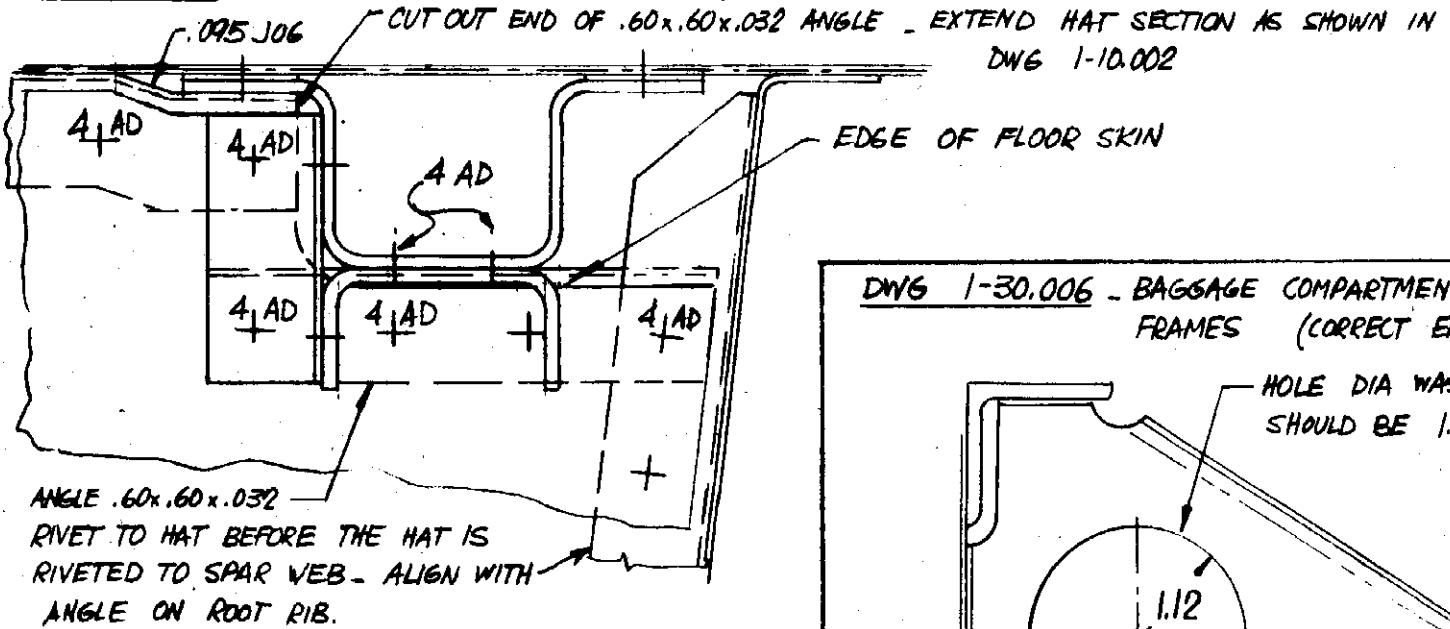
DRAWING 1-30.008 - COCKPIT DETAILS - FUSELAGE - (MANDATORY CHANGE)

IN ORDER TO INCREASE THE 5/16 BEARING AREA, MAKE THE FOLLOWING CHANGES

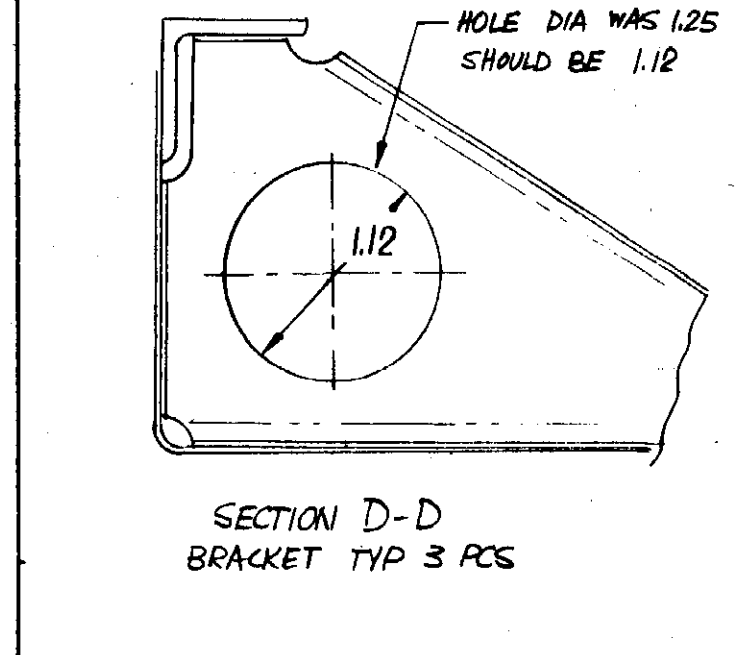
- 1) - INCREASE THICKNESS OF DOUBLERS FROM .063 TO .080
- 2) - REDUCE LENGTH OF SPACER TUBE FROM 1.85 TO 1.82
- 3) - INCREASE LENGTH OF AN 175 BOLT AS SHOWN.



DWG 1-10.008 - BL 9.60 RIB - (MANDATORY CHANGE)



DWG 1-30.006 - BAGGAGE COMPARTMENT FRAMES (CORRECT ERROR)

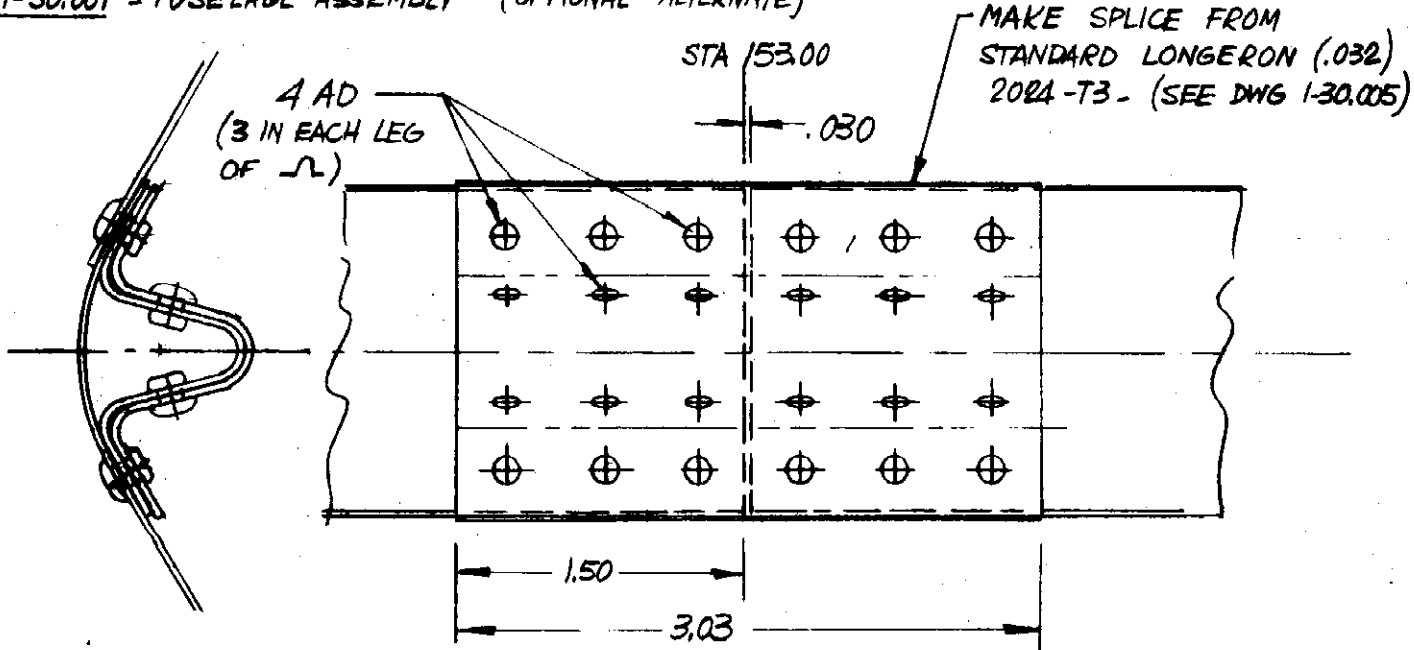


ENGINEERING CHANGE NOTICE # 2

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SHEET # 3

DWG 1-30.001 - FUSELAGE ASSEMBLY (OPTIONAL ALTERNATE)

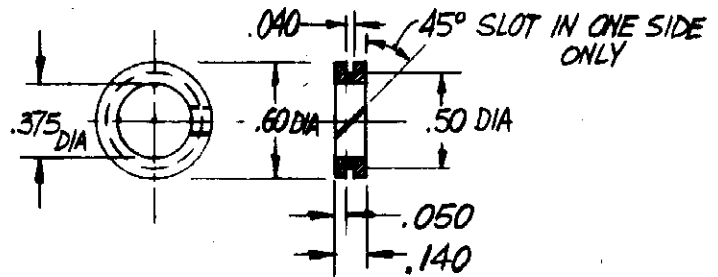
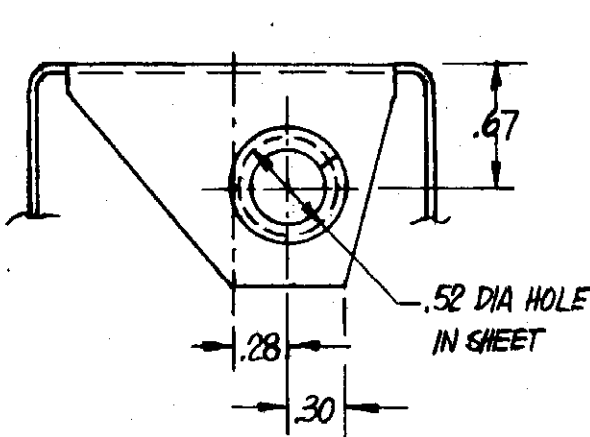


THE FUSELAGE TAIL CONE LONGERONS MAYBE SPLICED AS SHOWN AT STA 153.00 - THIS WILL PERMIT THE BENDING OF \angle IN A 6 FOOT BRAKE - CHANGE MATERIAL FROM "O" TO T3 (HEAT TREATED)

DWG 1-30.001 - FUSELAGE ASSEMBLY - (CORRECT ERROR. SEE DWG 1-50.008)

CHANGE THICKNESS OF BOTTOM SKIN BETWEEN STA 43.0 AND STA 57.23 (2) FROM .032 TO .040.

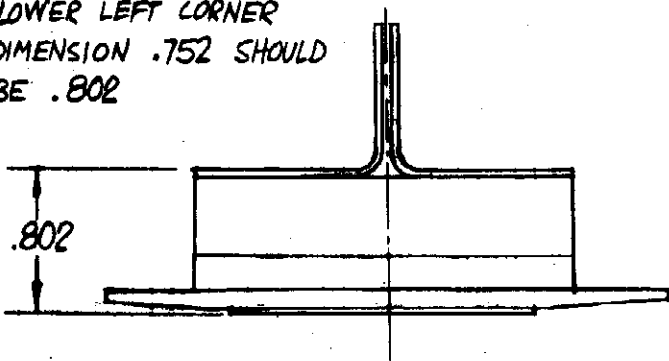
DWG 1-50.004 - FLAP LEVER CONTROLS - (CORRECT ERROR)



DETAIL OF NYLON GROMMET. ALTERNATE TO SHAMBAN GROMMET CALLED OUT IN THE DWG. IF YOU SELECT TO BUY THE SHAMBAN GROMMET CHANGE TO PART NO. S-11154-6-A-Y THIS SAME GROMMET COULD BE USED FOR ROUTING THE 3/8 DIA FUEL LINE THROUGH WING RIBS

DWG 1-10.002 - MAIN SPAR WING. (CORRECT ERROR)

LOWER LEFT CORNER DIMENSION .752 SHOULD BE .802



LIST OF RAW MATERIAL - PAGE (2) - DWG 1-10.006 LEADING EDGE SPLICE - 2024-O. BARE .080 SHEET - 3.8x38 - (WAS 3.8x33)

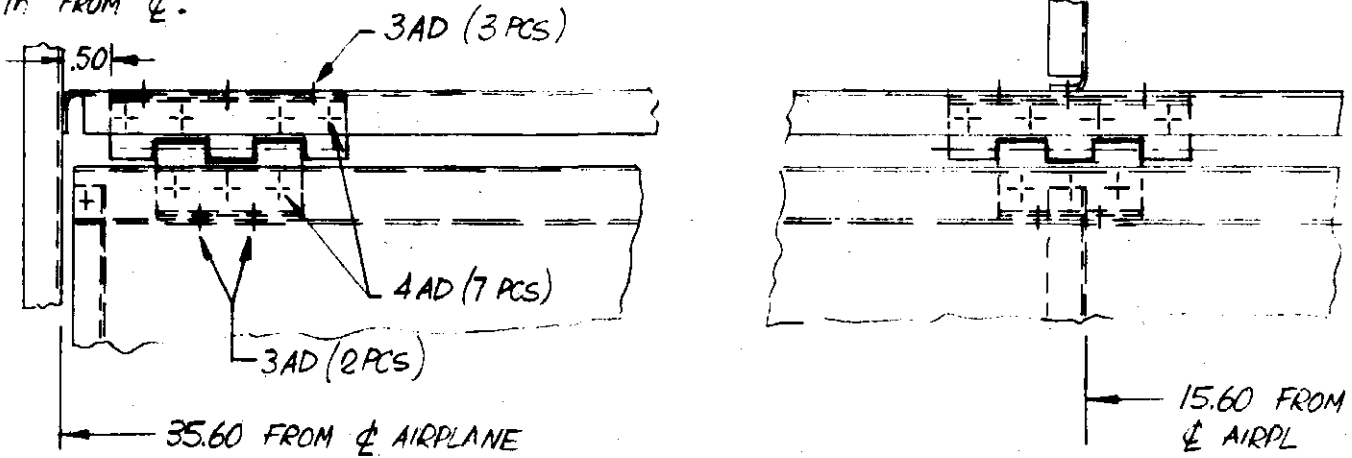
PAGE (10) - IN THE LAST .025 SHEET AT THE BOTTOM OF THE PAGE, CHANGE MATL. SIZE FOR FIN FRONT SPAR FROM 5.5x46 TO 5.5x48. AND FOR THE REAR SPAR FROM 5.5x44 TO 5.5x46.

ENGINEERING CHANGE NOTICE #2

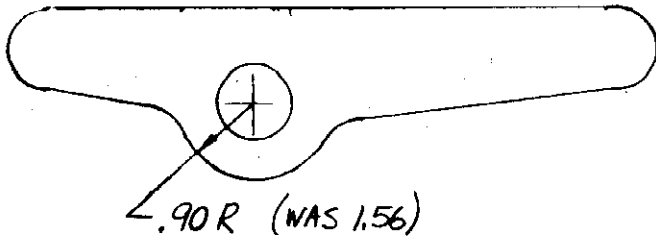
SHEET #4
NOV 1965

DWG 1-20.003 - STABILATOR - (MANDATORY CHANGE)

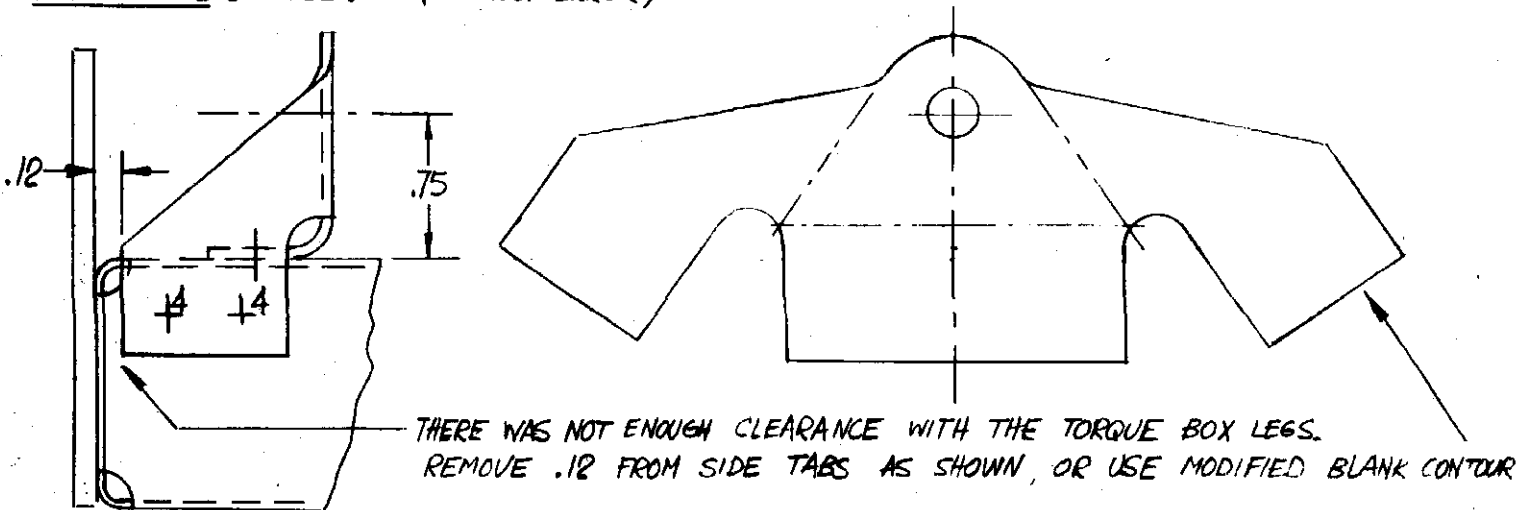
ADD HINGE AT THE OUTBD END OF TAB. RELOCATE EXISTENT HINGE AT 25.60 in FROM ϕ TO 15.60 in FROM ϕ .



DWG 1-50.002 - ALERON ROD - BELLCRANK & BALANCE - (CORRECT ERROR)



DWG 1-20.002 - RUDDER - (CORRECT ERROR)



DWG 1-40-002 - FUEL TANK ASSY - (DESIGN IMPROVEMENT TO FACILITATE FABRICATION. ELIMINATE THE EXTERNAL LAYERS OF 181 CLOTH ON TOP OF THE 1583 FOR THE ALTERNATE METHOD). CHANGE NOTES AS FOLLO:

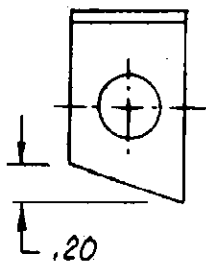
2. - - - - FOR -7 USE (2) LAYERS OF 1583 - DELETE: AND (1) LAYER (EXTERNAL) OF 181
FOR -9, -11, -25 & -29 USE (3) LAYERS OF 1583 - DELETE: ONE LAYER OF 181
 5. DELETE THE .0050 THICK INDUSTRIAL GLASS MAT BETWEEN MATTING SURFACES. IT PROVED OUT TO BE TOO DIFFICULT TO ASSEMBLE THE TANKS WITH THE MAT.
- ADD NOTE 9a - THE TANK SHOULD BE ASSEMBLED IN ONE OPERATION, SO ALL PARTS WILL FIT WITHOUT FORCING. SCOTCH TAPE COULD BE USED AS A PREASSEMBLY AID. TWO PERSONS ARE REQUIRED TO ASSEMBLY THE TANK IN ONE OPERATION. THE RESIN TAKES APPROX 1 1/2 HOUR TO SET. THE WHOLE ASSEMBLY SEQUENCE SHOULD BE REHEARSED BEFORE DOING THE "REAL" JOB. A WOOD CRADLE WILL BE VERY HELPFULL TO SUPPORT THE TANK

ENGINEERING CHANGE NOTICE # 2

SHEET # 5

NOV 1965

DWG 1-40.003 - FUEL TANK DETAILS - (CORRECT ERROR)



CUT OUT CORNER OF
-47 AS SHOWN TO
PROVIDE CLEARANCE
DURING ASSEMBLY

DIMENSIONS FOR CAMLOG KM 610-64 LATCH INSTALL.
SHOULD BE ONLY REFERENCE. USE THE PART TO DRILL
HOLES. BEND THE LEGS OF LATCH TO MATCH CONTOUR
LOCATE STRIKER -47 SO THE DOOR CLOSES SNUG.

DETAIL -47

DWG 1-50.006 - CONTROL DETAILS - (CORRECT ERROR)

ELEVATOR BEARING HOUSING - CALLS FOR TWO PARTS - SHOULD BE ONE LEFT AND ONE RIGHT

DWG 2-60.002 - NOSE LANDING GEAR ASSY - (CORRECT ERROR)

REM6-2N ROD END SHOULD BE RE3M6-2N.

ADD ANGLE OF NOSE WHEEL STEERING : SHOULD BE $\pm 25^\circ$ (THIS IS JUST A REFERENCE)

DWG 2-60.003 - LANDING GEAR DETAILS - (ALTERNATE METHOD - OPTIONAL CHANGE)

THE TUBE -35 COULD BE MACHINED ACCORDING TO THE FOLLOWING ALTERNATE METHOD.

1. WELD -37 AND -57 TO -35
2. CHECK FOR STRAIGHTNESS AND OUT OF ROUNDNESS - USING HYDRAULIC RAM STRAIGHTEN AS REQD.
3. CLEAN THE TUBE WITH EMERY CLOTH (240 GRIT)
4. DRILL .2500 DIA HOLE AT BOTTOM END (SHOWN IN DWGS. 2-60-001 AND 2-60-002)
5. MASK THE TUBE ENDS AND .2500 DIA HOLE - CHROME PLATE (THICKNESS .002 \pm .001)
6. PUT THE TUBE IN A LATHE AND HONE WITH A WET STONE (GRIT 400) OR EMERY CLOTH JUST TO ELIMINATE CHROME BUILD-UP. MOVE THE STONE SIDEWISE TO AVOID GROOVES.
7. MACHINE -29 LOWER BUSHING TO MATCH THE TUBE
8. JIG BORE -107 ADAPTER AND -39 ADAPTER TO MATCH .2500 DIA HOLE IN -35. IT IS VERY DIFFICULT TO DRILL THROUGH A CHROME PLATED PART.